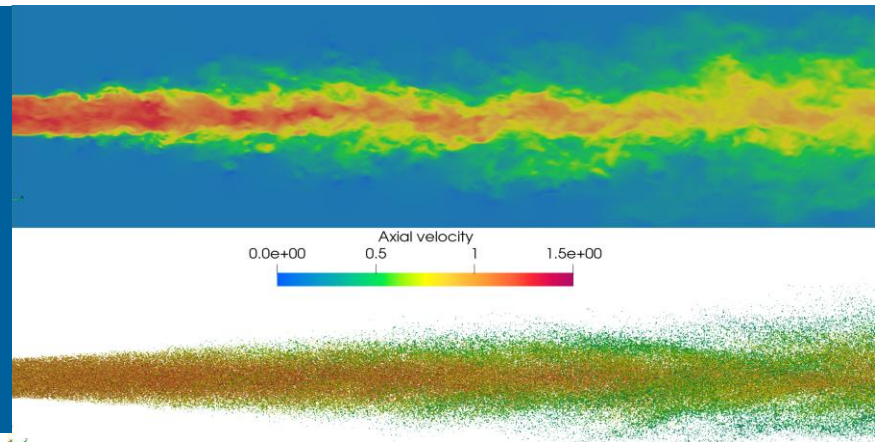


LARGE EDDY SIMULATION OF PARTICLE-LADEN FLOWS USING SPECTRAL ELEMENT METHOD



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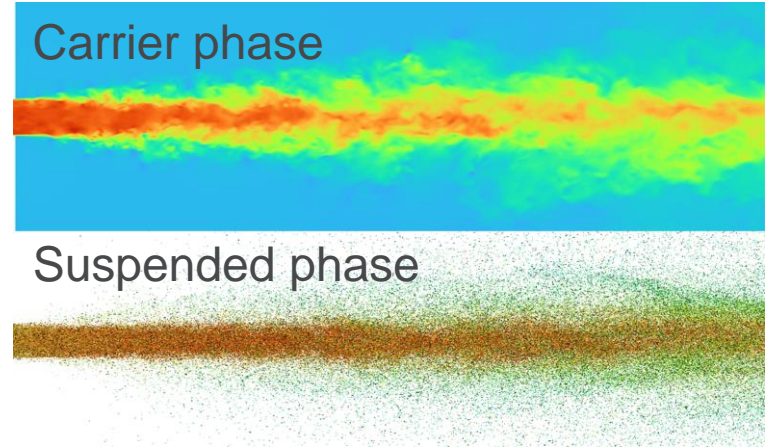
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OUTLINE

- Introduction
- Numerical method
- Computational setup
- Single-phase jet
- Particle-laden jet
 - **Effect of particle collisions**
 - **Effect of injection parameters**
 - Effect of sub-cycling, grid refinement, hydrodynamic forces, particle distribution
- Conclusions

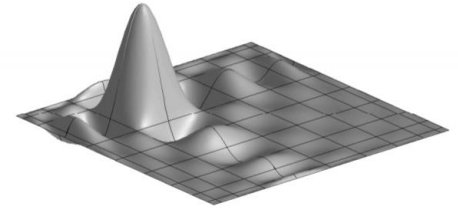
INTRODUCTION

- Turbulent particle-laden jet applications:
 - Gas-turbine engines, fluidized bed, flame spray pyrolysis, among others.
- Modeling approaches:
 - Eulerian-Eulerian (EE)
 - Eulerian-Lagrangian (EL)
- Levels of fidelity:
 - Reynolds-Averaged Navier-Stokes (RANS)
 - Large eddy simulation (LES)
 - Direct numerical simulation (DNS)/ Particle-resolved (PR) DNS
- Objective of this work:
 - Conduct high-fidelity **LES** of turbulent particle-laden jet flows using spectral-element method (**SEM**) in an **EL framework**.



NUMERICAL METHOD

- Spectral element method (SEM) (Patera, 1984; Maday & Patera, 1989) implemented in the Nek5000 code.
 - Weak formulation (Continuous-Galerkin)
 - N^{th} order tensor-product Lagrange polynomials at GLL points.
 $G = E(N + 1)^d$, G : grid points, E : elements
 - Exponential convergence with $N \rightarrow$ High accuracy at low cost.
 - Very low numerical dissipation and dispersion.
 - Low-Mach formulation for compressible flows.
 - Characteristics-based time integration \rightarrow CFL \sim 2.0
- Lagrangian stochastic parcels approach was used.
 - Particle ODEs are solved using RK3-SSP \rightarrow ppiclF library. (Zwick, 2019)
 - Spectral interpolation of gas phase solution is done at parcel locations.
 - Particle properties and source terms are projected on the Eulerian grid via a Gaussian projection filter.



2D basis function, $N=10$

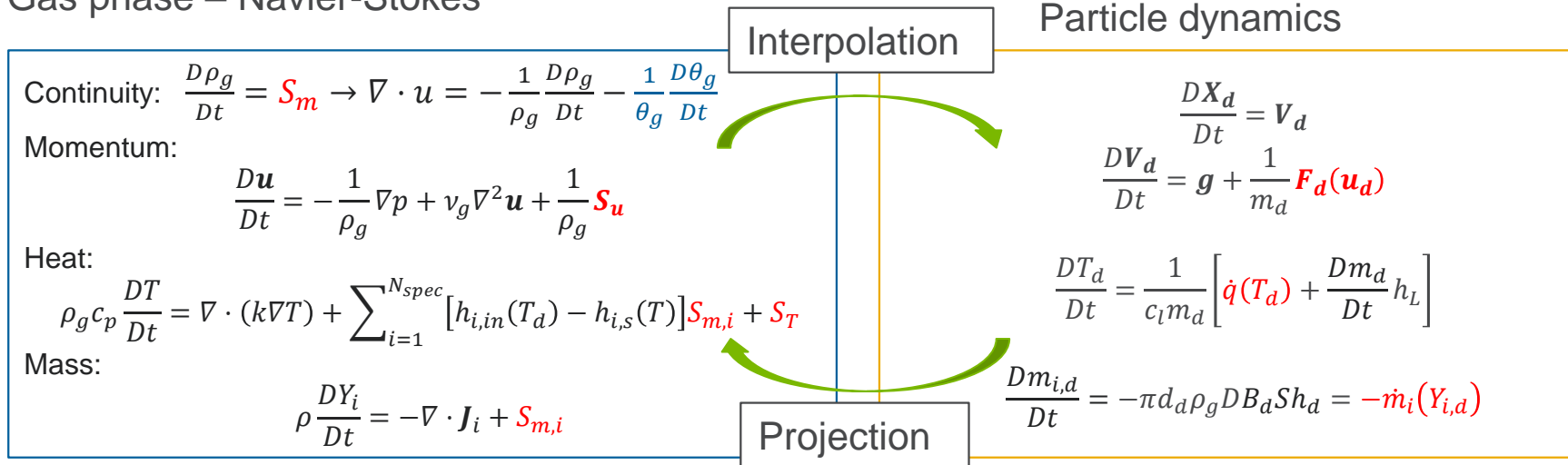
$$u_N(x) = \sum_{k=0}^N u_k h_k(x)$$

FLUID/PARTICLE COUPLING

$$\theta_l = 1 - \theta_g = \sum_{n=1}^{N_p} N_{d,n} \mathcal{V}G(|\mathbf{X}_n - \mathbf{x}|)$$

Gas phase – Navier-Stokes

Particle dynamics



$$S_u(\mathbf{x}) = \frac{1}{\theta_g} \sum_{n=1}^{N_p} -N_{d,n} \mathbf{F}_{d,n} G(|\mathbf{X}_n - \mathbf{x}|); \quad S_T(\mathbf{x}) = \frac{1}{\theta_g} \sum_{n=1}^{N_p} -N_{d,n} \dot{q}_n G(|\mathbf{X}_n - \mathbf{x}|); \quad S_m(\mathbf{x}) = \frac{1}{\theta_g} \sum_{n=1}^{N_p} N_{d,n} \dot{m}_n G(|\mathbf{X}_n - \mathbf{x}|)$$

N_p : number of parcels; N_d : number of droplets per parcel; θ_g : Void fraction

$$G(\mathbf{r}) = \left(\sqrt{\pi \delta_f^2 / 4 \ln 2} \right)^{-3} \exp \left[\frac{-|\mathbf{r}|^2}{\delta_f^2 / 4 \ln 2} \right]$$

Formulation from: Ling, Balachandar & Parmar (2016); Capecelatro & Desjardins (2013)

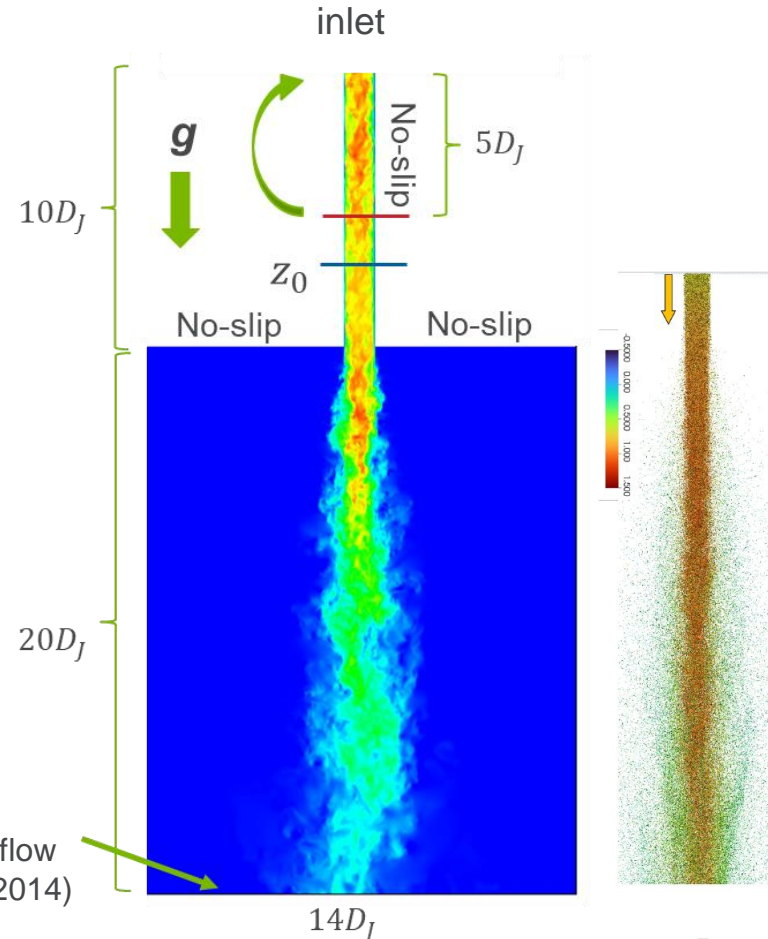
COMPUTATIONAL SETUP

Mostafa et al. (1989)

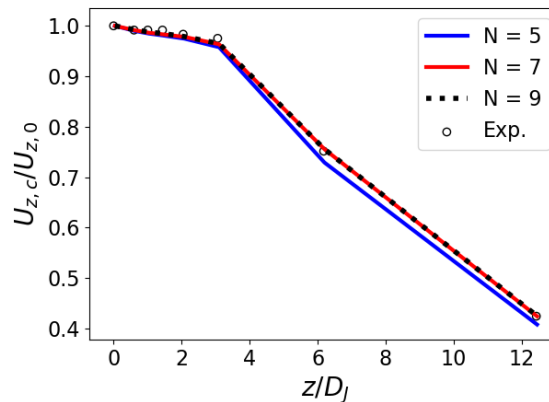
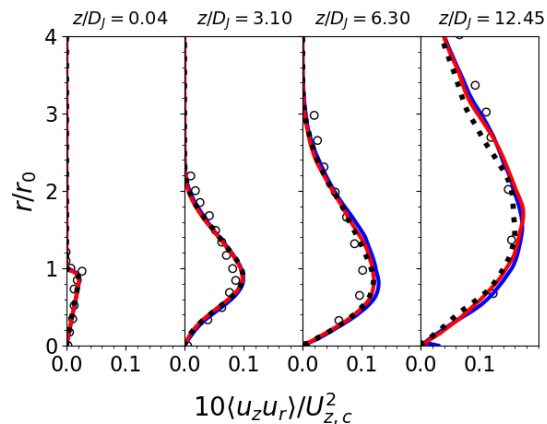
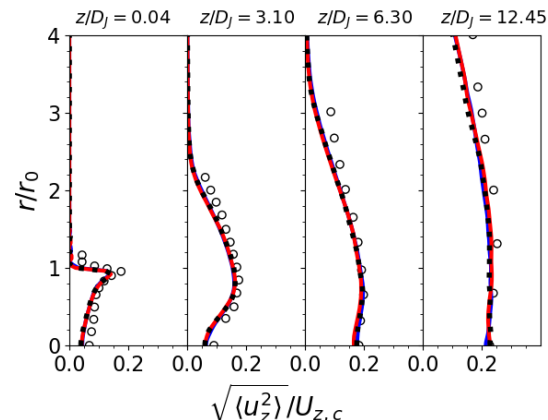
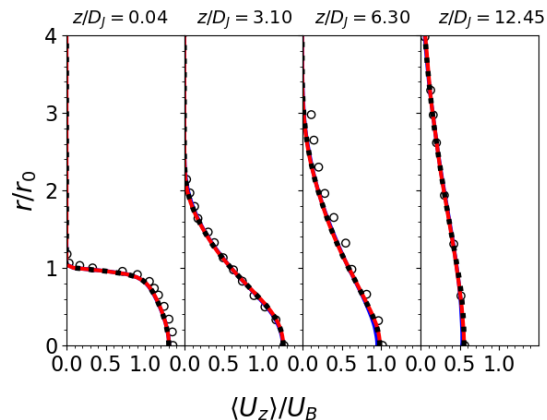
Parameter	Value
Gas	Air
Density	1.178 kg/m ³
Mass flow rate	0.0021 kg/s
Jet diameter, D_j	25.3 cm
Reynolds No.	5712
Bulk velocity, U_B	3.54 m/s
Particles	Glass
Density	2500 kg/m ³
Diameter	105 μm
Mass loading ratio, L	0.2, 1.0
Stokes No.	~11.6

$$L = \dot{m}_p / \dot{m}_g$$

- Initial particle velocity $V_0 / U_B = 0.5, 0.7, 0.9$
- Injection location $z_0 / D_j = -5.0, -3.7, -2.0$
- Realistic particle distribution based on experimental profiles
- $E = 75k$ elements
- $N = 7$
- $G = 25.8 M$
- $\frac{\Delta x_c}{D_j} = 0.0085 - 0.0471$

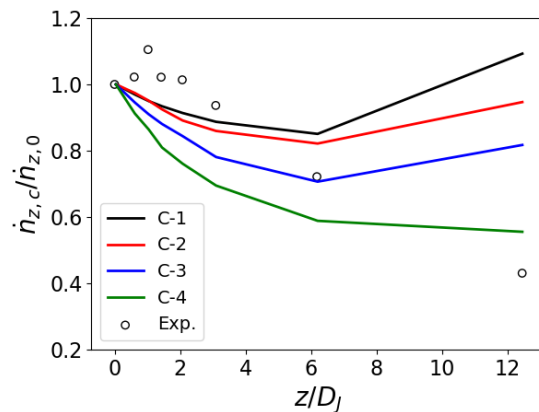
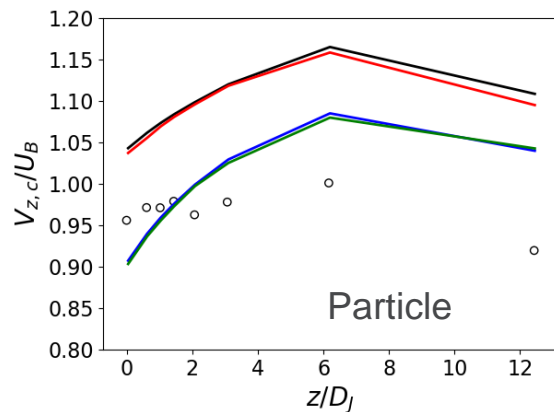
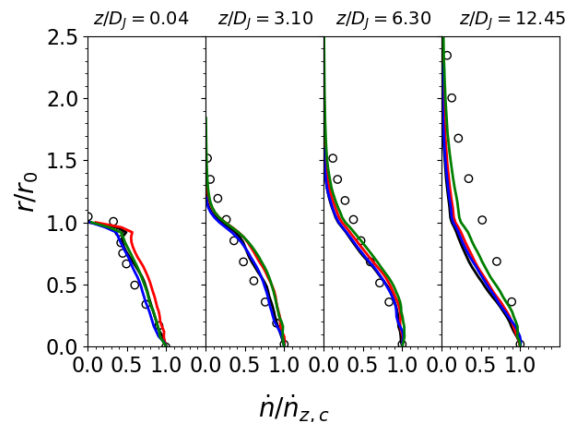
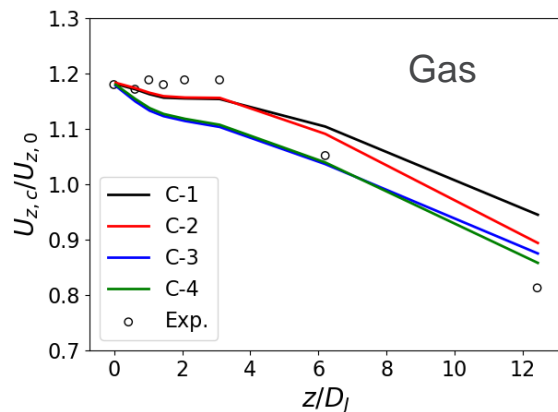


SINGLE-PHASE JET



Grid convergence

PARTICLE-LADEN JET: P-P COLLISIONS

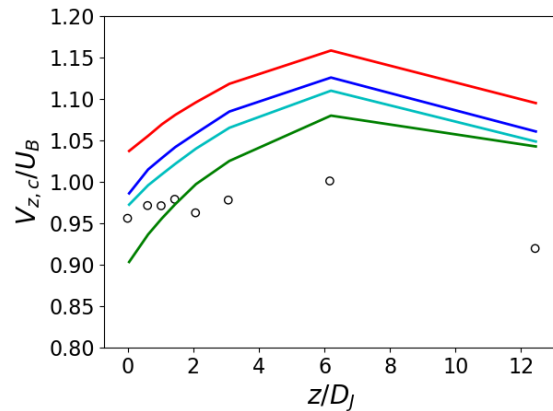
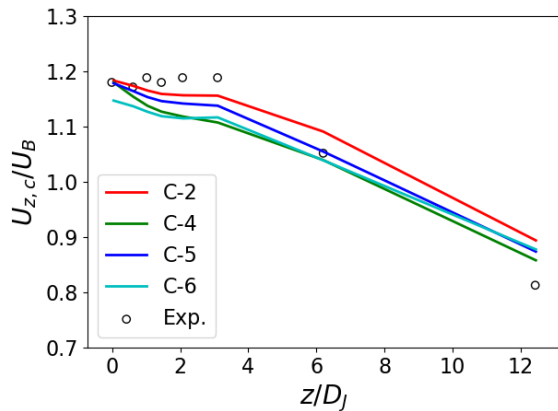
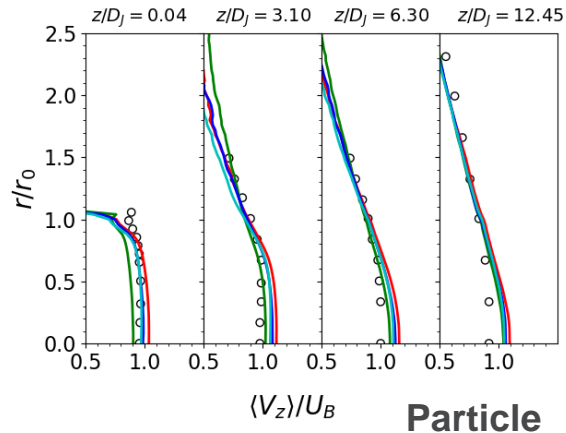
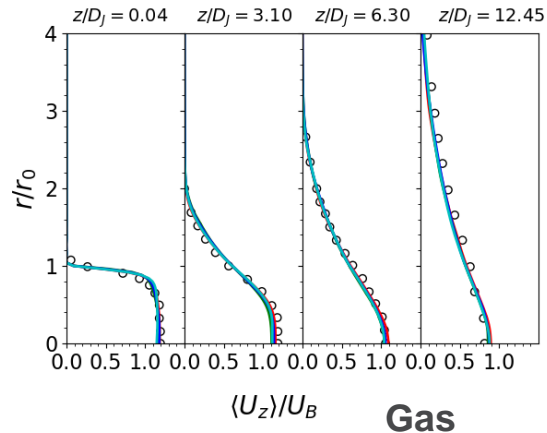


Case #	Injection loc., z_0/D_j	Coupling
C-1	-5	2-way
C-2	-5	4-way
C-3	-2	2-way
C-4	-2	4-way

$$L = 1.0; \bar{\theta}_p = 0.00047; St = 11.6$$

- Classification by Elghobashi (1991): limit between dense/dilute suspension. Collisions may be neglected.
- Current results: 4-way coupling improved particle distribution and centerline velocities.

PARTICLE-LADEN JET: INJECTION PARAMS.



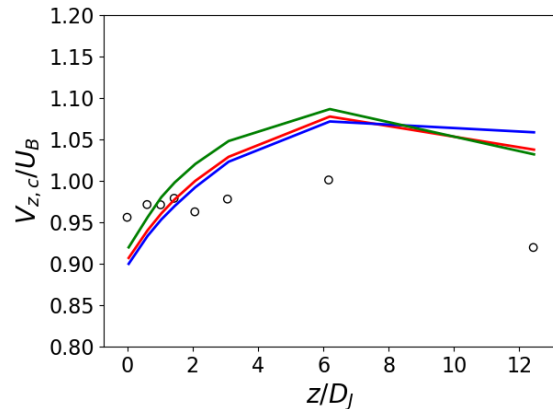
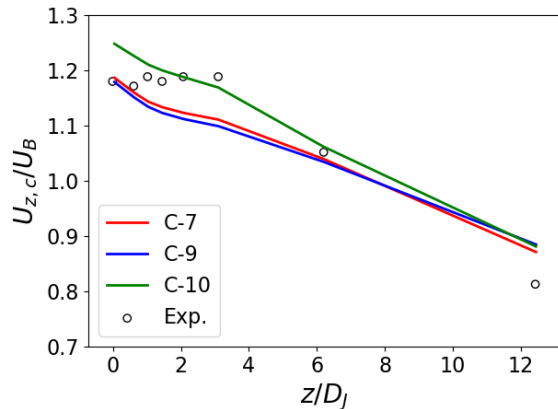
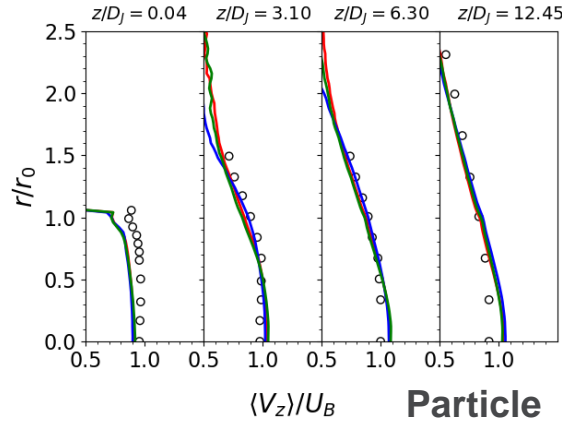
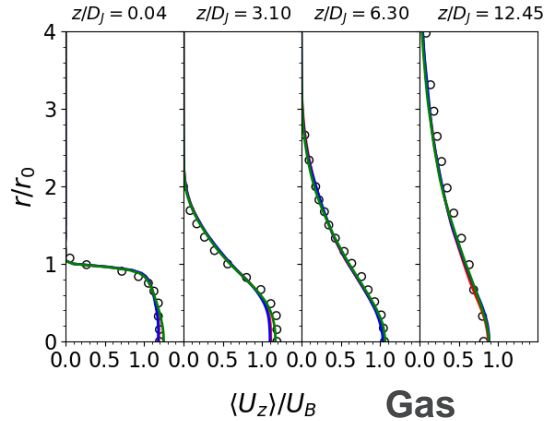
Case #	Injection loc., z_0/D_j	Injection vel. V_0/U_B
C-2	-5	0.7
C-4	-2	0.7
C-5	-3.7	0.7
C-6	-5	0.5

$$L = 1.0; \bar{\theta}_p = 0.00047; St = 11.6$$

$z_0 \downarrow$: Further upstream

→ By changing particle injection location and velocity, predicted exit velocity can be improved.

PARTICLE-LADEN JET: HYDRODYNAMIC FORCES AND PARTICLE DISTRIBUTIONS

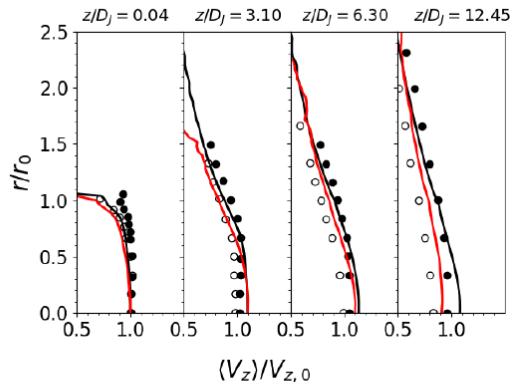


Case #	Forces	Distribution
C-7	Drag	Non-uniform
C-9	All HD	Non-uniform
C-10	Drag	Uniform

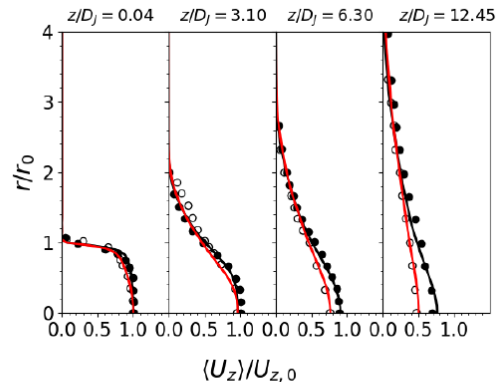
$$L = 1.0; \bar{\theta}_p = 0.00047; St = 11.6$$

→ Only drag plays a significant role, with other forces (pressure-gradient, added mass, shear-induced lift) having negligible effect on the mean flow.

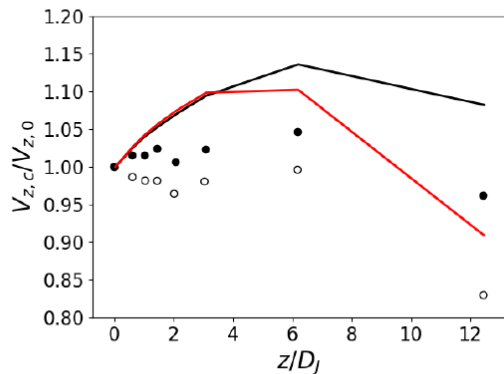
PARTICLE-LADEN JET: MASS LOADING



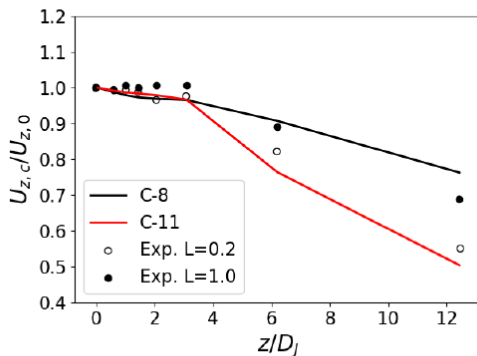
(a)



(b)



(c)



(d)

$$L = 0.2 - 1; \bar{\theta}_p = 9.5 \times 10^{-5}; St = 11.6$$

- At lower loading ratio (L), particles and gas phase tend to decelerate more downstream
- Current method can capture the difference in momentum transfer for cases with differing mass loading ratios, with numerical results showing the same trends as in the experiment.

PARTICLE-LADEN JET

- Further analysis not shown in this presentation:
 - Particle sub-cycling: did not have significant effect on flow statistics.
 - Grid sensitivity: Improving grid resolution by doing p-refinement (increasing N) did not affect results → gas-phase is well resolved.
 - Initial particle distribution: Assuming uniformly-distributed particles only affected the near-field, while far-field mean flow statistics are not very much affected by initial particle distribution.

Colmenares, J. D., Ameen, M. M., Wu, S., & Patel, S. (2021). Large Eddy Simulation of Turbulent Particle-laden Jets using the Spectral Element Method. In *AIAA Scitech 2021 Forum* (p. 0635).

CONCLUSIONS

- Current EL-SEM approach was used successfully to model turbulent single-phase and particle-laden jets.
- Varying particle injection location and velocity helped improve flow prediction.
- Collisions affected the flow, other simulation parameters did not.
- Cause for discrepancy between numerical and experimental results is unclear:
 - Missing information about the experimental flow:
 - Prior to exiting the inlet pipe (e.g. swirl)
 - Within the jet
 - Missing forces in the model:
 - History forces
 - Tangential component of collision forces
 - Realistic collisions (stiffness and restitution coefficients)
- Future work will include missing physics. More detailed experiments would help.

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THANK YOU!

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